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## INTRODUCTION

- ❖ Exposure to environmental noise in urban areas has direct or indirect effects on the health and well-being of populations according to the World Health Organization (WHO).
- ❖ Sound absorption solutions are proposed to **reduce the noise footprint** of an Electrically Distributed Anti Torque (EDAT) helicopter tail rotor.
- ❖ A multi-scale numerical modeling approach is being developed for designing sound absorption solutions to guide the design of an **optimal acoustic solution**.



## OBJECTIVES

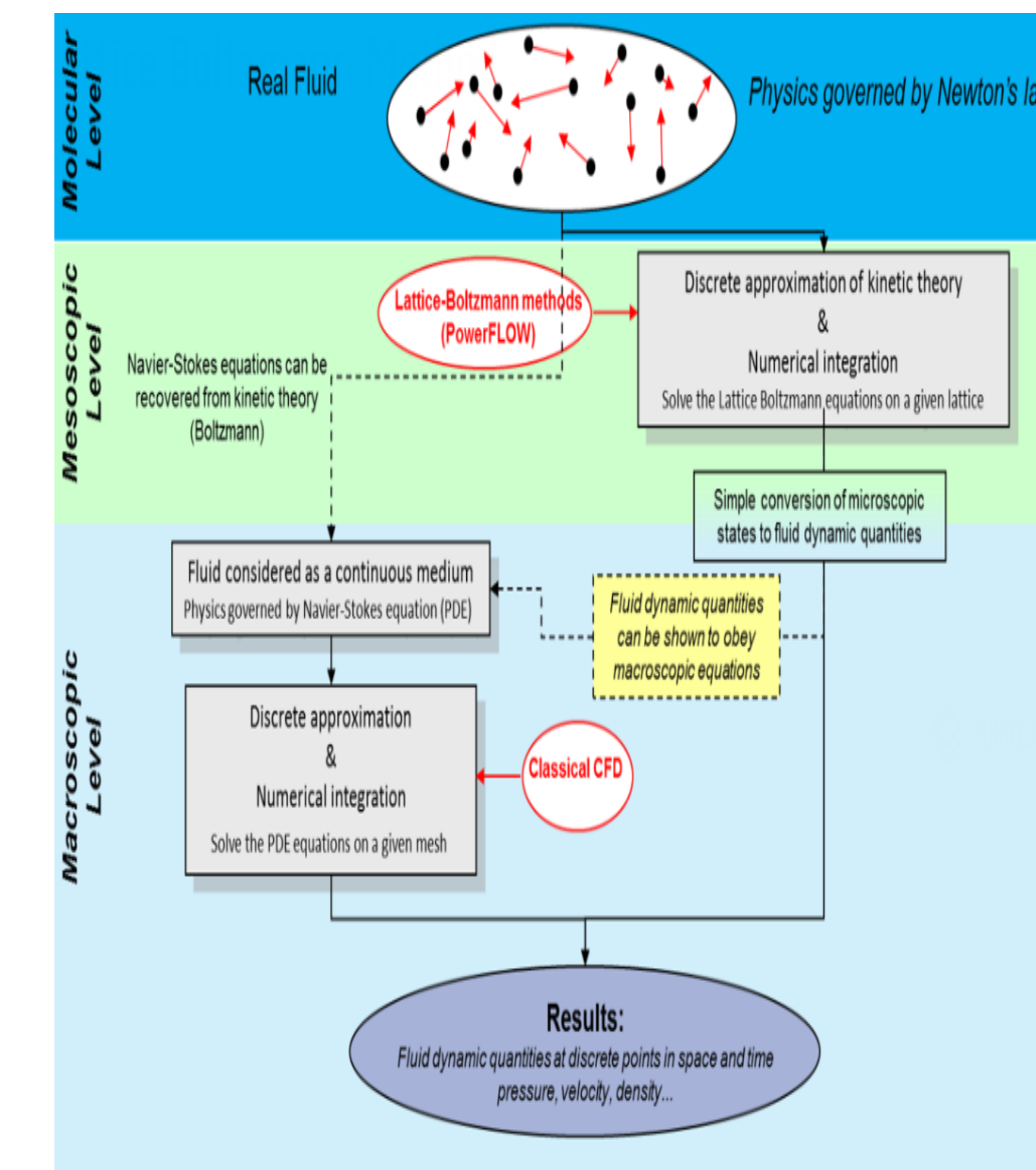
**General objective:**

- ✓ Establish a general research methodology to develop sound absorption solutions to reduce the noise footprint of shrouded rotors.

**Specific objectives:**

- ✓ Development of a numerical model for simulating the noise footprint of a shrouded rotor.
- ✓ Design and optimization of innovative sound absorption solutions using a multi-scale approach.
- ✓ Validation of the noise reduction obtained with the developed solution.

## THE LATTICE BOLTZMANN METHOD (LBM)



**Consider the Boltzmann equation in the form:**

$$\begin{cases} \partial_t f + \vec{\xi} \cdot (\nabla_x f) = \Omega \\ \Omega^{(BGK)} = \frac{f^{(eq)} - f}{\tau} \end{cases}$$

**Discretization**

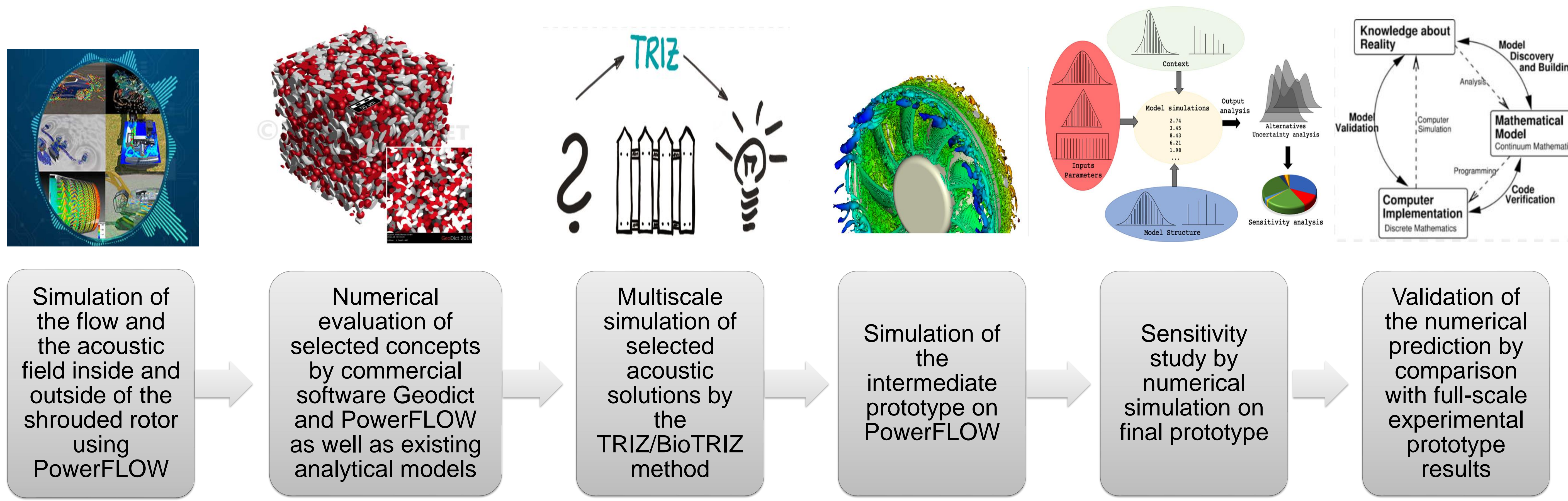
- ▶ Space and velocity space
- $\partial_t f_i + \vec{\xi}_i \cdot (\nabla_x f_i) = \Omega_i$

**Using a Taylor expansion**

$$\left( \frac{\partial f_i}{\partial t} + \vec{\xi}_i \cdot \frac{\partial f_i}{\partial \vec{x}} \right) = \frac{f_i(\vec{x} + \vec{\xi}_i \Delta t, t + \Delta t) - f_i(\vec{x}, t)}{\Delta t} + O(|\Delta \vec{x}|, \Delta t)$$

- ▶ The discrete Boltzmann equation is then  $f_i(\vec{x} + \vec{\xi}_i \Delta t, t + \Delta t) - f_i(\vec{x}, t) = \Omega_i \Delta t + O((\Delta \vec{x})^2, (\Delta t)^2)$

## METHODOLOGY

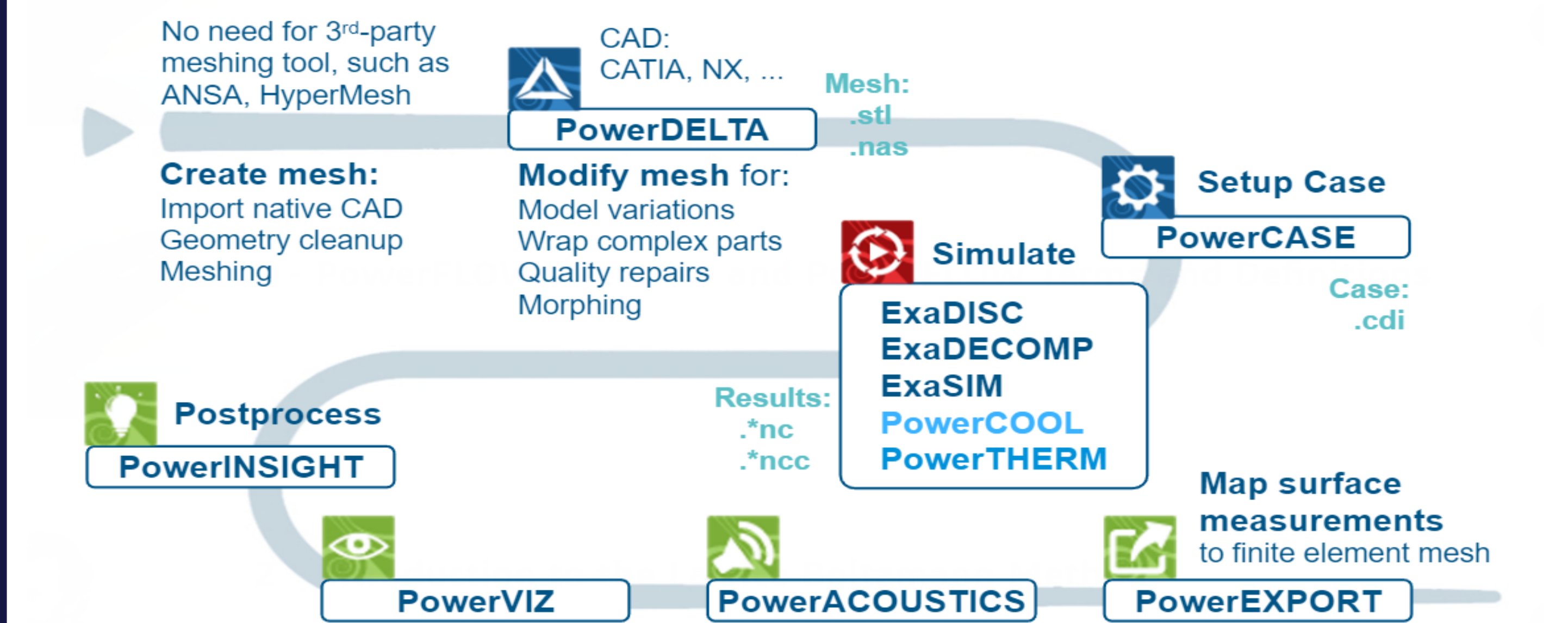


## POWERFLOW SIMULATION

**Three pillars of PowerFLOW**

- ❖ Aerodynamics Simulations
- ❖ Thermal Simulations
- ❖ Aeroacoustics Simulations

**Application Usage Flow**



## CONCLUSION AND UPCOMING STEPS

This project aims at demonstrating that it is possible to implement a design methodology to achieve a substantial overall noise reduction on a single rotor.

- Analytical models available in the literature and PowerFLOW numerical models will be used to predict the acoustic response of resonators and other acoustic solutions.
- To test the acoustic solutions before their fabrication, a campaign of multi-scale numerical simulations will be held.
- Multi-scale numerical simulation with PowerFLOW will allow us to validate the measurements of acoustic pressure fields in the near and far field with and without the addition of acoustic treatment.

## NOMENCLATURE

- $f$ : Particle distribution function
- $\xi$ : Microscopic particle velocity
- $t$ : Time
- $\tau$ : Relaxation time
- $\Omega$ : BGK collision operator
- $f^{eq}$ : Distribution function at equilibrium

## ACKNOWLEDGMENTS

