

## Introduction

A series of 3D numerical investigations of the incompressible flow over a 30P30N high-lift airfoil with a span of 20% of stowed chord length,  $c$ , are performed at low Reynolds number ( $Re_c \sim O(10^4)$ ) at an angle of attack  $\alpha = 4^\circ$ . A well-resolved direct numerical simulation is implemented via NEK5000[1] which is an open-source code based upon a high-order spectral element method. The present study focuses on  $Re_c = 0.832 \times 10^4$  and  $1.270 \times 10^4$ , aiming to understand the instability mechanisms that fundamentally govern the transition route to turbulence over the high-lift airfoil.

## Flow configuration

- A total of 332,050 elements
- a 7<sup>th</sup>-polynomial order in space & 3<sup>rd</sup>-order accuracy in time

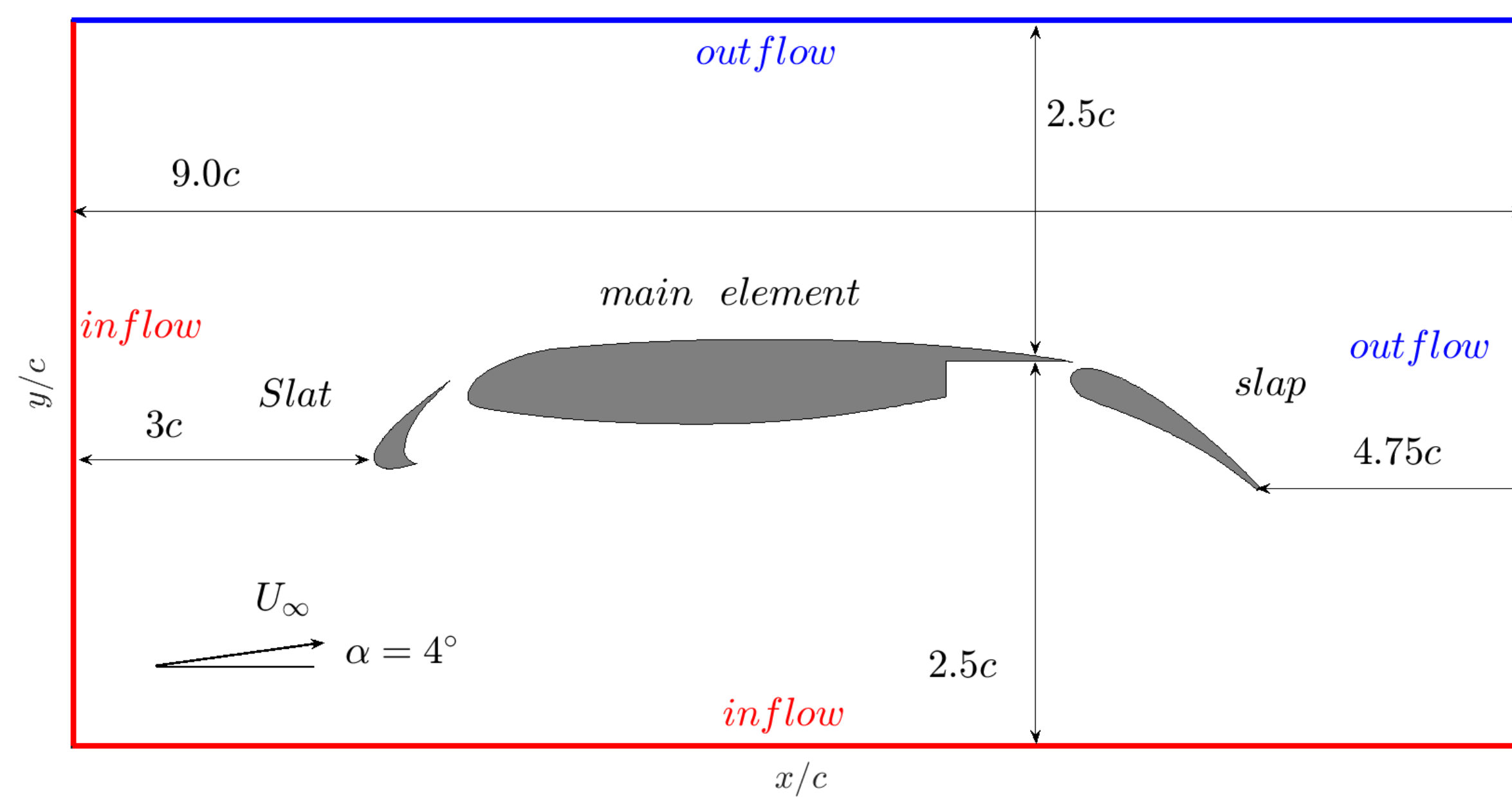


Figure 1. Sketch of the computational domain used for the calculation.

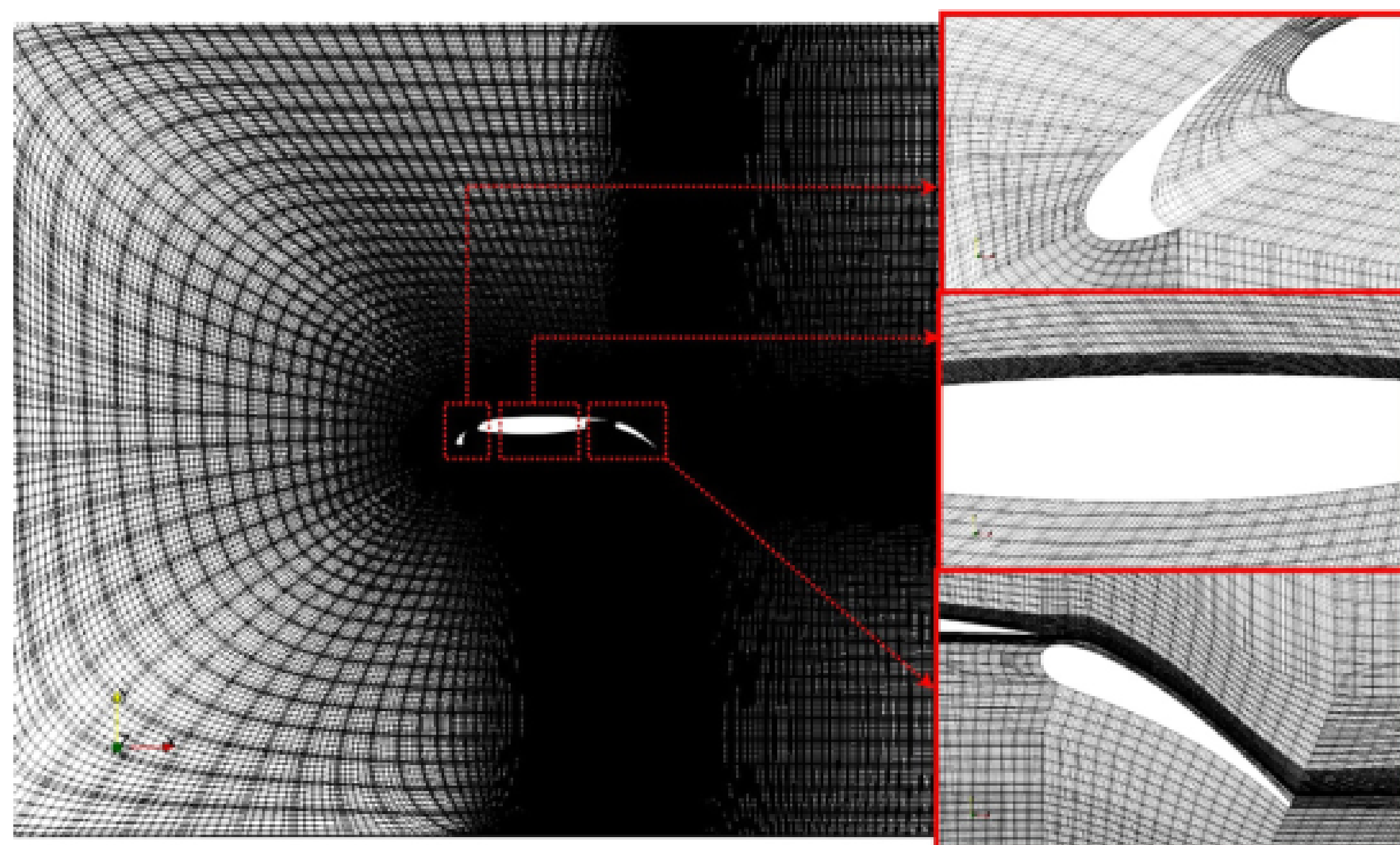


Figure 2. Grid mesh around the airfoil [2].

## Coherent structures

Counter-rotating Görtler vortices, characteristic of centrifugal instability, are observed in the streamwise direction shortly downstream of the main-airfoil-element leading edge for  $Re_c = 0.832 \times 10^4$  and  $1.27 \times 10^4$ . This phenomenon, however, is unexpected over a convex surface. The appearance of Görtler vortices is likely caused by a curved shear layer emanating from the slat cove, just above the leading edge of the main-airfoil-element, as posited by Wang et al. [3]. The present work looks into this phenomenon more closely. As the flow evolves further downstream and starts to separate, Kelvin-Helmholtz (K-H) instability takes over and leads to the formation of quasi two-dimensional (2D) vortices. While propagating downstream, these quasi-2D coherent structures get distorted and reoriented towards the streamwise direction prior to the occurrence of more complex structures in the separation region. **One question this study attempts to answer** is whether the Görtler vortices extend sufficiently far downstream such that they start to interact with the K-H instability.

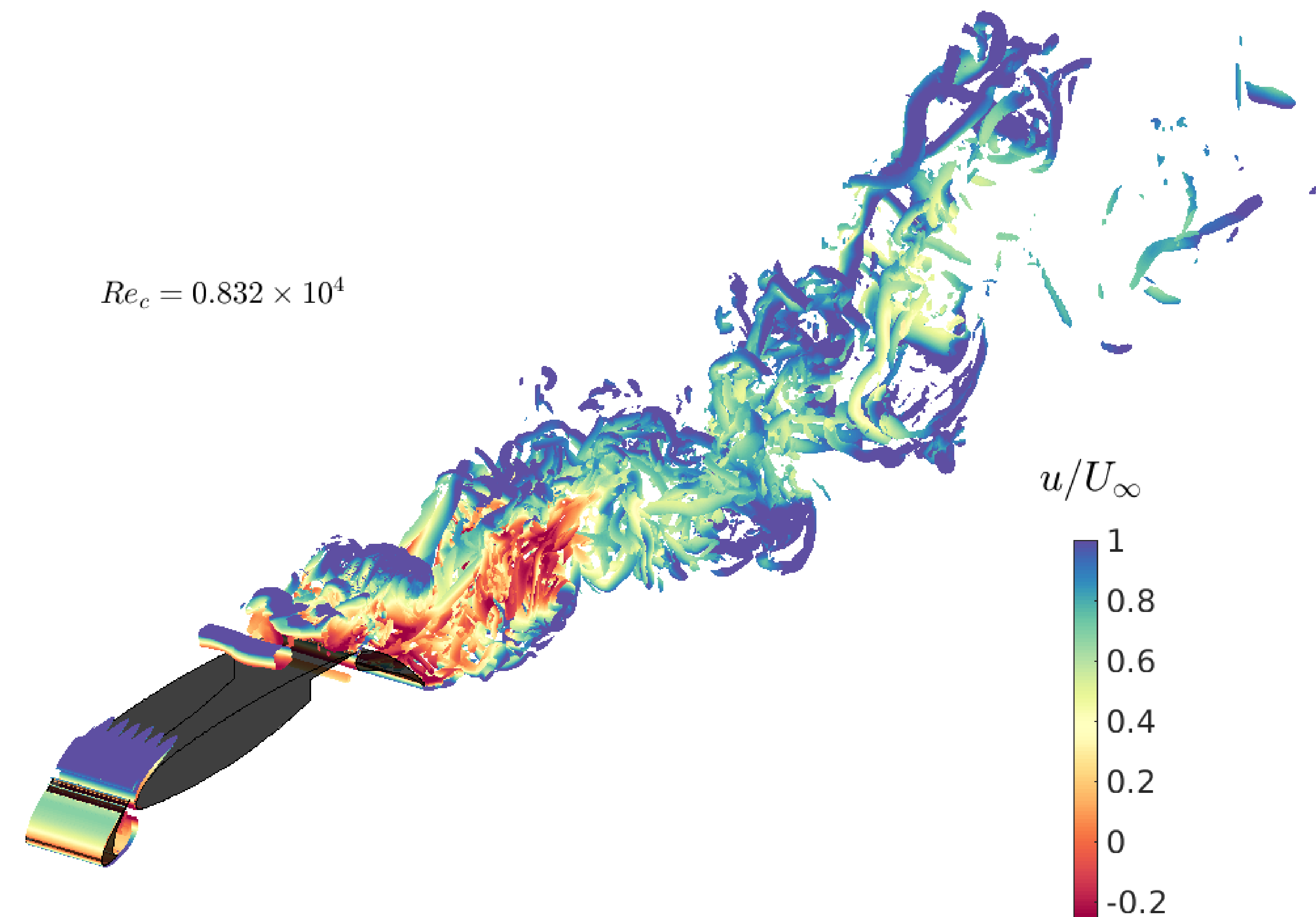


Figure 3. Iso-surfaces of  $\lambda_2$  ( $\lambda_2 = -50.0$ ), coloured by the magnitude of streamwise velocity, at  $t/c = 8.96$

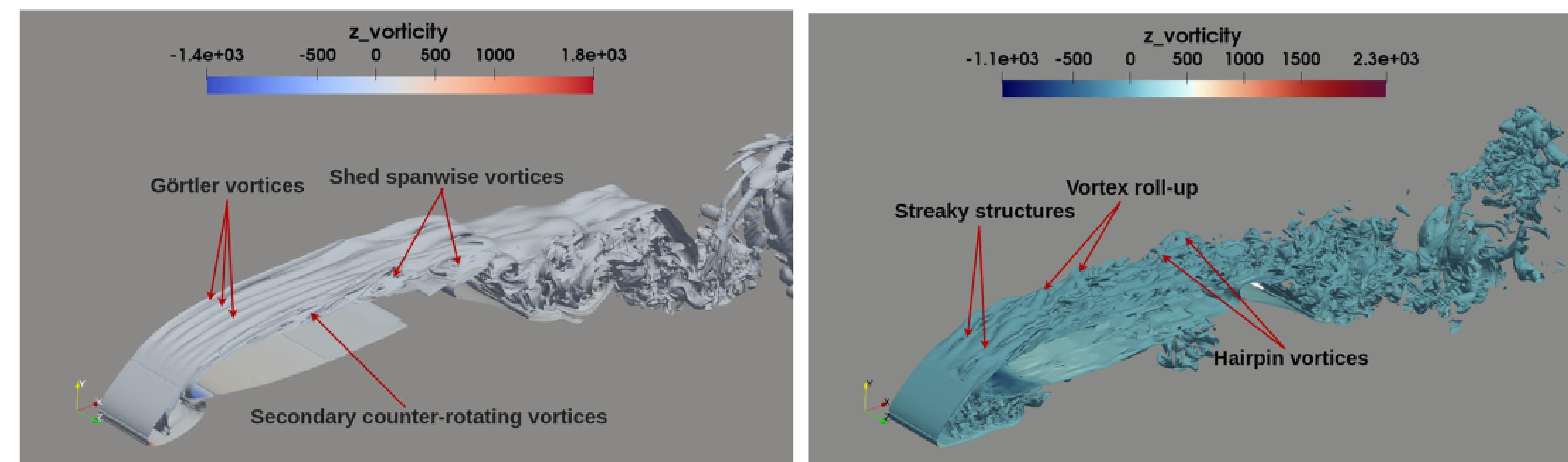


Figure 4. Instantaneous spanwise vorticity for (LEFT)  $Re_c = 1.27 \times 10^4$  and (RIGHT)  $Re_c = 1.83 \times 10^4$  [2]

## References

- [1] P. F. Fischer, J. W. Lottes, and S. G. Kerkemeier. NEK5000. <https://nek5000.mcs.anl.gov>, 2008.
- [2] M Vadsola, GG Agbaglah, and C Mavriplis. Slat cove dynamics of low reynolds number flow past a 30p30n high lift configuration. *Physics of Fluids*, 33(3):033607, 2021.
- [3] Jiang-Sheng Wang, Li-Hao Feng, Jin-jun Wang, and Tian Li. Görtler vortices in low-reynolds-number flow over multi-element airfoil. *Journal of Fluid Mechanics*, 835:898–935, 2018.